# Table of Contents

Table of Contents.............................................................................................................. i  
Section 1 ...................................................................................................................... ... 1  
   Introduction .................................................................................................................. 1  
      1.1 Overview ............................................................................................................. 1  
      1.2 Naming Conventions........................................................................................... 1  
      1.3 Features .............................................................................................................. 2  
      1.4 Technical Specifications ...................................................................................... 3  
         1.4.1 Physical Characteristics ................................................................................ 3  
         1.4.2 Environmental Characteristics ...................................................................... 3  
         1.4.3 Electrical Characteristics ............................................................................... 3  
         1.4.4 Performance.................................................................................................. 3  
         1.4.5 Interfaces ...................................................................................................... 4  
      1.5 Application .......................................................................................................... 5  
         1.5.1 Application Considerations............................................................................ 5  
Section 2 ...................................................................................................................... ... 7  
   Operational Characteristics.......................................................................................... 7  
      2.1 Self Test.............................................................................................................. 7  
      2.2 Initialization ......................................................................................................... 7  
      2.3 Navigation ........................................................................................................... 8  
      2.4 Satellite Data Collection ...................................................................................... 8  
Section 3 ...................................................................................................................... .1 0  
   Hardware Interface..................................................................................................... 10  
      3.1 Mechanical Dimensions .................................................................................... 10  
      3.2 Connector Specifications .................................................................................. 10  
      3.3 Connector Pin-Out ............................................................................................ 11  
      3.4 Antenna Connection ......................................................................................... 14  
Section 4 ...................................................................................................................... .1 5  
   Software Interface ...................................................................................................... 15  
      4.1 NMEA Received sentences .............................................................................. 16  
         4.1.1 Almanac Information (ALM) ........................................................................ 16  
         4.1.2 Sensor Initialization Information (PGRMI) ................................................... 17  
         4.1.3 Sensor Configuration Information (PGRMC)............................................... 17  
         4.1.4 Output Sentence Enable/Disable (PGRMO) ............................................... 18  
      4.2 NMEA Transmitted Sentences.......................................................................... 19  
         4.2.1 Sentence Transmission Rate ...................................................................... 19  
         4.2.2 Transmitted Time ........................................................................................ 20  
         4.2.3 Global Positioning System Almanac Data (ALM) ........................................ 21  
         4.2.4 Global Positioning System Fix Data (GGA)................................................. 21  
         4.2.5 GPS DOP and Active Satellites (GSA) ....................................................... 22  
         4.2.6 GPS Satellites in View (GSV) ..................................................................... 22  
         4.2.7 Recommended Minimum Specific GPS/TRANSIT Data (RMC).............. 22  
         4.2.8 Track Made Good and Ground Speed with GPS Talker ID (VTG).............. 23  

4.2.9 Geographic Position with LORAN Talker ID (LCGLL) ........................................ 23
4.2.10 Track Made Good and Ground Speed with LORAN Talker ID (LCVTG) ... 23
4.2.11 Estimated Error Information (PGRME) .......................................................... 23
4.2.12 GPS Fix Data Sentence (PGRMF) ................................................................. 24
4.2.13 Sensor Status Information (PGRMT) ............................................................. 24
4.2.14 3D velocity Information (PGRMV) .............................................................. 25
4.3 Baud Rate Selection ......................................................................................... 25
4.4 One-Pulse-Per-Second Output ....................................................................... 25
4.5 RTCM Received Data ....................................................................................... 26
Appendix A ............................................................................................................. 27
  Earth Datums ........................................................................................................ 27
Appendix B ............................................................................................................. 30
  GPS 25LP Connectors .......................................................................................... 30
Appendix C ............................................................................................................. 31
  GPS 25LP Evaluation Kit ...................................................................................... 31
Section 1

Introduction

1.1 Overview

The GARMIN GPS 25LP Series are GPS sensor boards designed for a broad spectrum of OEM (Original Equipment Manufacturer) system applications. The GPS 25LPs will simultaneously track up to twelve satellites providing fast time-to-first-fix, one second navigation updates and low power consumption. Their far-reaching capability meets the sensitivity requirements of land navigation as well as the dynamics requirements of high performance aircraft.

The GPS 25LP design utilizes the latest surface mount technology as well as high-level circuit integration to achieve superior performance while minimizing space and power requirements. All critical components of the system including the RF/IF receiver hardware and the digital baseband are designed and manufactured by GARMIN to ensure the quality and capability of the GPS 25LP sensor board. This hardware capability combined with software intelligence makes the board set easy to integrate and use.

The GPS 25LP is designed to withstand rugged operating conditions, however it should be mounted in an enclosure as part of a larger system designed by an OEM or system integrator. A minimum system must provide the sensor board with conditioned input power and L1 GPS RF signal. The system may communicate with the board set via a choice of two CMOS/TTL or two RS-232 compatible bi-directional communications channels. A highly accurate one-pulse-per-second (PPS) output can be utilized in applications requiring precise timing measurements. An on-board memory rechargeable backup battery allows the sensor board to retain critical data such as satellite orbital parameters, last position, date and time. Non-volatile memory is also used to retain board configuration settings even if backup battery power fails. End user interfaces such as keyboards and displays are added by the application designer.

1.2 Naming Conventions

The GPS 25LP sensors are delineated with a three letter extension to designate the operating voltage range and the serial data voltage specification.

High Voltage - GPS25-HVx designation indicates that the unit will accept a high input voltage. The internal switching regulator will operate from a 6VDC to 40VDC unregulated supply.
Low Voltage - GPS25-LVx designation indicates that the unit is designed to operate from a low voltage 3.6VDC to 6.0VDC supply. Operation at about 4VDC is the most power efficient mode of operation for the GPS25LP sensor. The unit is protected if a high voltage is inadvertently applied to the input.

RS-232 Serial Data - GPS25-xVS designation means that the two bi-directional serial data ports are true RS-232 ports conforming to the RS-232E standard.

CMOS Serial Data - GPS25-xVC designation means that the two bi-directional serial data ports use CMOS output buffers. The inputs buffers will accept either CMOS(TTL) voltage levels or RS-232 voltage levels. This configuration is adequate for communicating directly with serial devices over short cable lengths (less than 20 meters).

1.3 Features

The GPS 25LP sensor boards provide a host of features that make it easy to integrate and use.

1) Full navigation accuracy provided by Standard Positioning Service (SPS)
2) Compact design ideal for applications with minimal space
3) High performance receiver tracks up to 12 satellites while providing fast first fix and low power consumption
4) Differential capability utilizes real-time RTCM corrections producing less than 5 meter position accuracy
5) On-board clock and memory are sustained by a rechargeable memory backup battery which recharges during normal operation or by optional external standby power
6) User initialization is not required.
7) Two communication channels and user selectable baud rates allow maximum interface capability and flexibility. The standard channels are CMOS/TTL levels for the -xVC version or RS-232 for the xVS versions.
8) Highly accurate one-pulse-per-second output for precise timing measurements. The default pulse width is 100 msec, however it is configurable in 20 msec increments from 20 msec to 980 msec.
9) Flexible input voltage levels of 3.6Vdc to 6.0Vdc with overvoltage protection in the -LVx, and 6.0Vdc to 40Vdc in the -HVx versions.
10) Fully shielded construction for maximum EMI and RFI protection
11) FLASH based program memory. New software revisions upgradeable through serial interface

1.4 Technical Specifications

Specifications are subject to change without notice.

1.4.1 Physical Characteristics

1) Single board integrated with complete component shielding
2) Weight: 1.3 ounce [38g]
3) Size: 1.83” (w) x 2.75” (l) x 0.45” (h)

1.4.2 Environmental Characteristics

1) Operating temperature: -30°C to +85°C (board temperature)
2) Storage temperature: -40°C to +90°C

1.4.3 Electrical Characteristics

1) Input voltage: +3.6VDC to 6.0VDC regulated, 150 mVp-p ripple -LVx versions. +6.0VDC to 40VDC unregulated -HVx version.
2) Input current: 140 mA typical 150 mA max -LVx versions, 20 mA while in power down.
   1000mW typical 1200mW max -HVx version, 300μA while in power down.
3) Backup power: 3V Rechargeable Lithium cell battery, up to 6 month charge
4) Receiver sensitivity: -165dBW minimum

1.4.4 Performance

1) Tracks up to 12 satellites (up to 11 with PPS active)
2) Update rate: 1 second
3) Acquisition time
   - 15 seconds warm (all data known)
   - 45 seconds cold (initial position, time and almanac known, ephemeris unknown)
   - 1.5 minutes AutoLocate™ (almanac known, initial position and time unknown)
   - 5 minutes search the sky (no data known)
4) Position accuracy:
   Differential GPS (DGPS): Less than 5 meters RMS
   Non-differential GPS:  15 meters RMS (100 meters with Selective Availability on)

5) Velocity accuracy: 0.2 m/s RMS steady state (subject to Selective Availability)

6) Dynamics:  999 knots velocity, 6g dynamics

7) One-pulse-per-second accuracy: ±1 microsecond at rising edge of PPS pulse
   (subject to Selective Availability)

1.4.5 Interfaces

1) Dual channel CMOS/TTL level (-xVC versions) or RS-232 compatible level (-xVS
   versions), with user selectable baud rate (300, 600,1200, 2400, 4800, 9600, 19200)

2) NMEA 0183 Version 2.0 ASCII output (GPALM, GPGGA, GPGSA, GPGSV,
   GPRMC, GPVTG, PGRME, PGRMT, PGRMV, PGRMF, LCGLL, LCVTG)
   Inputs
   - Initial position, date and time (not required)
   - Earth datum and differential mode configuration command, PPS Enable,
     almanac
   Outputs
   - Position, velocity and time
   - Receiver and satellite status
   - Differential Reference Station ID and RTCM Data age
   - Geometry and error estimates

3) Real-time Differential Correction input (RTCM SC-104 message types 1,2,3 and 9)

4) One-pulse-per-second timing output

5) Optional binary TTL output data format
1.5 Application

1.5.1 Application Considerations

1) The GPS 25LP sensor boards contain a sensitive receiver. Additional electromagnetic shielding may be required to prevent undesirable interference from other nearby circuits.
2) The GPS 25LP sensor boards use approximately 0.5 W to .85 W, depending on supply voltage, and require minimal cooling. Forced air cooling is not recommended since it may cause rapid temperature changes which may temporarily affect the frequency stability of the internal oscillator.

3) Interruptions in the RF signal can increase acquisition time. Antenna location with clear line-of-sight to all directions in the sky will yield the best performance.

4) The GPGSV sentence contains signal strength information for the visible satellites. Typical values will be between 30 db and 50 db. A majority of values near the lower limit may indicate a marginal RF signal.
Section 2

Operational Characteristics

This section describes the basic operational characteristics of the GPS 25LP sensor boards. Additional information regarding input and output specifications are contained in Section 4.

2.1 Self Test

After input power has been applied to the GPS 25LP sensor boards and periodically thereafter, the units will perform critical self test functions and report the results over the output channel(s). The following tests will be performed:

1) RAM check
2) FLASH test
3) Receiver test
4) Real-time clock test
5) Oscillator check

In addition to the results of the above tests, the board set will report software version information.

2.2 Initialization

After the initial self test is complete, the GPS 25LP will begin the process of satellite acquisition and tracking. The acquisition process is fully automatic and, under normal circumstances, will take approximately 45 seconds to achieve a position fix (15 seconds if ephemeris data is known). After a position fix has been calculated, valid position, velocity and time information will be transmitted over the output channel(s).

Like all GPS receivers, the GPS 25LP utilizes initial data such as last stored position, date and time as well as satellite orbital data to achieve maximum acquisition performance. If significant inaccuracy exists in the initial data, or if the orbital data is obsolete, it may take 1.5 minutes to achieve a navigation solution. The GPS 25LP AutoLocate™ feature is capable of automatically determining a navigation solution without intervention from the host system. However, acquisition performance can be improved if the host system initializes the board set following the occurrence of one or more of the following events:
1) Transportation over distances further than 1500 kilometers
2) Failure of the on-board memory battery
3) Stored date/time off by more than 30 minutes

See Section 4 for more information on initializing the GPS 25LP.

2.3 Navigation

After the acquisition process is complete, the GPS 25LP will begin sending valid navigation information over its output channels. These data include:

1) Latitude/longitude/altitude
2) Velocity
3) Date/time
4) Error estimates
5) Satellite and receiver status

The GPS 25LP sensor boards will select the optimal navigation mode (2D or 3D) based on available satellites and geometry considerations. When navigating in the 2D mode the board set utilizes the last computed altitude or the last altitude supplied by the host system, whichever is newer. The host system must ensure that the altitude used for 2D navigation is accurate since the resulting position error may be as large as the altitude error. See Section 4 for information altitude initialization.

The GPS 25LP will default to automatic differential mode -- “looking” for real-time differential corrections in RTCM SC-104 standard format using message types 1,2,3 or 9 and attempting to apply them to the satellite data in order to produce a differential (DGPS) solution. The host system, at its option, may also command the board set to choose differential only mode. When navigating in the differential only mode, the board set will output a position only when a differential solution is available.

2.4 Satellite Data Collection

The GPS 25LP sensor boards will automatically update satellite orbital data as they operate. The intelligence of the board set combined with its hardware capability allows these data to be collected and stored without intervention from the host system. A few key points should be considered regarding this process:

1) If the sensor board is not operated for a period of six (6) months or more, the unit will “search the sky” in order to collect satellite orbital information. This process is fully automatic and, under normal circumstances, will take 3-4 minutes to achieve
a navigation solution. However, the host system should allow the board set to remain on for at least 5 minutes after the first satellite is acquired (see Section 4 for more information on status indications).

2) If the memory backup battery fails, the sensor board will search the sky as described above. The system designer should be aware of the availability of standby power input to the board set to prevent this situation.

3) If the initial data is significantly inaccurate, the board set will perform an operation known as AutoLocate™. This procedure is fully automatic and, under normal circumstances, will require 1.5 minutes to calculate a navigation solution. AutoLocate™, unlike search the sky, does not require that the sensor board continue to operate after a fix has been obtained.
Section 3

Hardware Interface

3.1 Mechanical Dimensions

The GPS 25LP sensor board mounts via four #4 or M-3 size screws (see drawing).

Top View                                      End View

3.2 Connector Specifications

The GPS 25LP sensor boards features a single row, right angle, 12-pin male connector. The mating connector is available from JST Corporation (see appendix B).
3.3 Connector Pin-Out

The following is a functional description of each pin of the interface connector. The pins are numbered from left to right as viewed from the connector end of the board with the connector on top.

Pin 1: TXD2 - Second Serial Asynchronous Output. Electrically identical to TXD1.

Pin 2: RXD2 - Second Serial Asynchronous Input. Electrically identical to RXD1. This input may be used to receive serial differential GPS data formatted per “RTCM Recommended Standards For Differential Navstar GPS Service, Version 2.1” (see Section 4 for more details).

Pin 3: PPS - One-Pulse-Per-Second Output. Typical voltage rise and fall times are 300 nSec. Impedance is 250 ohms. Open circuit output voltage is 0V and Vin. The default format is a 100 millisecond high pulse at a 1Hz rate, the pulse width is programmable from a configuration command in 20msec increments. Rising edge is synchronized to the start of each GPS second. This output will provide a nominal 700 mVp-p signal into a 50 Ohm load. The pulse time measured at the 50% voltage point will be about 50 nSec earlier with a 50 Ohm load than with no load.

Pin 4: TXD1 - First Serial Asynchronous Output. CMOS/TTL output levels vary between 0V and Vin in the -LVC version. In the -LVS and -HVS versions a RS-232 compatible output driver is available. This output normally provides serial data which is formatted per “NMEA 0183, Version 2.0”. Switchable to 300, 600, 1200, 2400, 4800, 9600 or 19200 BAUD. The default BAUD is 4800. This output data functions in parallel with pin 12.
Pin 5: RXD1 - First Serial Asynchronous Input. RS-232 compatible with maximum input voltage range \(-25 < V < 25\). This input may be directly connected to standard 3 to 5Vdc CMOS logic. The minimum low signal voltage requirement is 0.8V, and the maximum high signal voltage requirement is 2.4V. Maximum load impedance is 4.7K ohms. This input may be used to receive serial initialization/configuration data, as specified in Section 4.1.

Pin 6: POWER DOWN - External Power Down Input. Inactive if not connected or less than 0.5V. Active if greater than 2.7V. Typical switch point is 2.0V @ 0.34 mA. Input impedance is 15K Ohms. Activation of this input powers the internal regulators off and drops the supply current below 20mA. The computer will be reset when power is restored.

Pin 7: VAUX - Auxiliary External Backup Battery Charge circuit. If used, a 4VDC to 33VDC power source is required to supply a trickle charge to the battery. During normal operation a trickle charge is supplied to the battery from an internal regulator. On-board rechargeable battery capacity is 7 mA hour.

Pin 8: GND - Power and Signal Ground

Pin 9: NC - Do NOT connect this pin on the GPS 25LP.

Pin 10: Vin - Regulated +3.6V to +6V, 200 mA (maximum) in the -LVx versions. Typical operating current is 140 mA. Transients and overvoltages are protected by an internal 6.8V transient zener diode and a positive temperature coefficient thermistor. With voltages greater than 6.8Vdc the thermistor will eventually power the unit off until proper supply voltages are returned. Antenna supply is derived from Vin after passing through a 50mA current limiter. The CMOS/TTL output buffers are powered by Vin, therefore a 3.6Vdc supply will create 3.6V logic output levels.

In the -HVS version, Vin can be an unregulated 6.0Vdc to 40Vdc. This voltage drives a switching regulator with a nominal 4.4Vdc output, which powers the antenna connector current limiter, the internal linear regulators, and the CMOS output buffers.

Pin 11: NC - Do NOT connect this pin on GPS 25LP.

Pin 12: NMEA - NMEA 0183, Version 1.5 electrical specification compatible serial output. This output is CMOS compatible with a no load voltage swing of 0.2Vdc to 0.9xVCC. This output normally provides ASCII sentences formatted per “NMEA 0183, Version 2.0”. User selectable baud rates of 300, 600, 1200, 2400, 4800, 9600 and 19200 are available, with 4800 as the default. The data output on this pin is identical to the data output on pin 4.
**Typical Embedded Application Interconnect**

For embedded applications where the sensor board will be connected directly to the host system microprocessor through standard CMOS logic, the following connections are suggested.

![Diagram of GPS25-LVC Sensor Board connections](image)

**Typical Personal Computer Application Interconnect**

The following connections are suggestions for applications where the sensor board will be connected to a personal computer via a standard RS-232 interface.

![Diagram of GPS25-LVS Sensor Board connections](image)
3.4 Antenna Connection

The GPS 25LP sensor boards make their antenna connections via a 50 ohm MCX style connector attached directly to the sensor board (+Vin-0.4V @ 15 mA power is supplied on the center conductor for the antenna).

Coaxial cable supplies +Vin-0.4V @15 mA to antenna/preamp. 1.57542 GHz signal returns to the sensor board.

The recommended GARMIN remote antenna for the GPS 25LP series is the GA27. It has eight feet of RG-174A/U type cable with a MCX connector installed for attachment to the sensor board. An extension cable with up to -5dB additional attenuation at 1.5 GHz may be used with the GA27 (either 50 Ohm or 75 Ohm impedance cable is permitted).

A passive antenna may be used with the GPS 25LP series sensor boards, provided no more than -2dB or -3dB cable loss is inserted between the antenna and the sensor.

Higher gain competitors antennas may be used provided there is adequate cable attenuation to prevent overloading the sensitive GPS 25LP receiver. 15dB excessive antenna system gain (LNA Gain - Cable Loss) should be considered a maximum. Check to insure that the antenna in question will operate properly with the antenna bias voltage that is supplied by the GPS 25LP, when powered by the desired Vin voltage. The GPS 25LP will current limit for loads above 50mA to prevent damage, should the antenna cable become accidentally shorted.

The following table may be used to estimate the typical attenuation of various cable types at 1.5 GHz. For example, a 15 foot length of RG-188 may have 4.5 dB
attenuation. The attenuation of different styles of the same cable type may vary. Refer to the specifications of the actual cable that will be used.

<table>
<thead>
<tr>
<th>Type</th>
<th>Typical Attenuation dB/100’ @ 1 GHz</th>
</tr>
</thead>
<tbody>
<tr>
<td>RG-211A/228A</td>
<td>4.5</td>
</tr>
<tr>
<td>RG-217/224/293/14</td>
<td>5.8</td>
</tr>
<tr>
<td>RG-8/215/216</td>
<td>9</td>
</tr>
<tr>
<td>RG-6</td>
<td>11</td>
</tr>
<tr>
<td>RG-59</td>
<td>11.5</td>
</tr>
<tr>
<td>RG-58</td>
<td>20</td>
</tr>
<tr>
<td>RG-188</td>
<td>30</td>
</tr>
<tr>
<td>RG-174</td>
<td>31</td>
</tr>
<tr>
<td>RG-178</td>
<td>45</td>
</tr>
</tbody>
</table>
Section 4

Software Interface

The GPS 25LP interface protocol design is based on the National Marine Electronics Association’s NMEA 0183 ASCII interface specification, which is fully defined in “NMEA 0183, Version 2.0” (copies may be obtained from NMEA, P.O. Box 50040, Mobile, AL, 36605, U.S.A.) and the Radio Technical Commission for Maritime Services’ “RTCM Recommended Standards For Differential Navstar GPS Service, Version 2.1, RTCM Special Committee No. 104” (copies may be obtained from RTCM, P.O. Box 19087, Washington, D.C., 20036, U.S.A.). The GPS 25LP interface protocol, in addition to transmitting navigation information as defined by NMEA 0183, transmits additional information using the convention of GARMIN proprietary sentences.

The following sections describe the data format of each sentence transmitted and received by the GPS 25LP sensor board. The baud rate selection, one-pulse-per-second output interfaces and RTCM differential GPS input are also described.

4.1 NMEA Received sentences

The subsequent paragraphs define the sentences which can be received on RXD1 by the GPS 25LP sensor boards. Null fields in the configuration sentence indicate no change in the particular configuration parameter.

All sentences received by the GPS 25LP must be terminated with <CR><LF>, but do not require the checksum *hh. The checksum is used for parity checking data and it is recommended that the checksum be used in environments containing high electromagnetic noise. It is generally not required in normal PC environments. Sentences may be truncated by <CR><LF> after any data field and valid fields up to that point will be acted on by the GPS 25LP.

4.1.1 Almanac Information (ALM)

$GPALM,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>,<10>,<11>,<12>,<13>,<14>,<15> *hh<CR><LF>

The $GPALM sentence can be used to initialize the sensor board’s stored almanac information if battery back-up has failed.

<1> Total number of ALM sentences to be transmitted by the sensor board during almanac download. This field can be null or any number when sending almanac to the sensor board.
<2> Number of current ALM sentence. This field can be null or any number when sending almanac to the sensor board.

<3> Satellite PRN number, 01 to 32.

<4> GPS week number.

<5> SV health, bits 17-24 of each almanac page.

<6> Eccentricity

<7> Almanac reference time.

<8> Inclination angle.

<9> Rate of right ascension.

<10> Root of semi major axis.

<11> Omega, argument of perigee.

<12> Longitude of ascension node.

<13> Mean anomaly

<14> afo clock parameter

<15> af1 clock parameter

4.1.2 Sensor Initialization Information (PGRMI)

The $PGRMI sentence provides information used to initialize the sensor board set position and time used for satellite acquisition. Receipt of this sentence by the board set causes the software to restart the satellite acquisition process. If there are no errors in the sentence, it will be echoed upon receipt. If an error is detected, the echoed PGRMI sentence will contain the current default values. Current PGRMI defaults can also be obtained by sending $PGRMIE, to the board.

$PGRMI,<1>,<2>,<3>,<4>,<5>,<6>,<7>*hh<CR><LF>

<1> Latitude, ddmm.mmm format (leading zeros must be transmitted)

<2> Latitude hemisphere, N or S

<3> Longitude, dddmm.mmm format (leading zeros must be transmitted)

<4> Longitude hemisphere, E or W

<5> Current UTC date, ddmmyy format

<6> Current UTC time, hmmss format

<7> Receiver Command, A = Auto Locate, R = Unit Reset.

4.1.3 Sensor Configuration Information (PGRMC)

The $PGRMC sentence provides information used to configure the sensor board operation. Configuration parameters are stored in non-volatile memory and retained between power cycles. The GPS 25LP will echo this sentence upon its receipt if no errors are detected. If an error is detected, the echoed PGRMC sentence will contain the current default values. Current default values can also be obtained by sending $PGRMCE, to the board.
$PGRMC,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>,<10>,<11>,<12>,<13>,<14>*hh<CR><LF>

<1> Fix mode, A = automatic, 2 = 2D exclusively (host system must supply altitude), 3 = 3D exclusively
<2> Altitude above/below mean sea level, -1500.0 to 18000.0 meters
<3> Earth datum index. If the user datum index (96) is specified, fields <4> through <8> must contain valid values. Otherwise, fields <4> through <8> must be null. Refer to Appendix A for a list of earth datums and the corresponding earth datum index.
<4> User earth datum semi-major axis, 6360000.0 to 6380000.0 meters (.001 meters resolution)
<5> User earth datum inverse flattening factor, 285.0 to 310.0 (10^9 resolution)
<6> User earth datum delta x earth centered coordinate, -5000.0 to 5000.0 meters (1 meter resolution)
<7> User earth datum delta y earth centered coordinate, -5000.0 to 5000.0 meters (1 meter resolution)
<8> User earth datum delta z earth centered coordinate, -5000.0 to 5000.0 meters (1 meter resolution)
<9> Differential mode, A = automatic (output DGPS data when available, non-DGPS otherwise), D = differential exclusively (output only differential fixes)
<10> NMEA Baud rate, 1 = 1200, 2 = 2400, 3 = 4800, 4 = 9600, 5 = 19200, 6 = 300, 7 = 600
<11> Velocity filter, 0 = No filter, 1 = Automatic filter, 2-255 = Filter time constant (10 = 10 second filter
<12> PPS mode, 1 = No PPS, 2 = 1 Hz
<13> PPS pulse length, 0-48 = (n+1)*20msec. Example n = 4 => 100 msec pulse
<14> Dead reckoning valid time 1-30 (sec)

All configuration changes take effect after receipt of a valid value except baud rate and PPS mode. Baud rate and PPS mode changes take effect on the next power cycle or an external reset event.

4.1.4 Output Sentence Enable/Disable (PGRMO)

The $PGRMO sentence provides the ability to enable and disable specific output sentences.

The following sentences are enabled at the factory: GPGSA, GPGSV, GPRMC, PGRME, PGRMT and PGRMV

$PGRMO,<1>,<2>*hh<CR><LF>

<1> Target sentence description (e.g., PGRMT, GPGSV, etc.)
Target sentence mode, where:

0 = disable specified sentence
1 = enable specified sentence
2 = disable all output sentences
3 = enable all output sentences (except GPALM)

The following notes apply to the PGRMO input sentence:

1) If the target sentence mode is ‘2’ (disable all) or ‘3’ (enable all), the target sentence description is not checked for validity. In this case, an empty field is allowed (e.g., $PGRMO,,3), or the mode field may contain from 1 to 5 characters.

2) If the target sentence mode is ‘0’ (disable) or ‘1’ (enable), the target sentence description field must be an identifier for one of the sentences being output by the GPS 25LP.

3) If either the target sentence mode field or the target sentence description field is not valid, the PGRMO sentence will have no effect.

4) $PGRMO,GPALM,1 will cause the sensor board to transmit all stored almanac information. All other NMEA sentence transmission will be temporarily suspended.

### 4.2 NMEA Transmitted Sentences

The subsequent paragraphs define the sentences which can be transmitted on TXD1 by the GPS 25LP sensor boards.

#### 4.2.1 Sentence Transmission Rate

Sentences are transmitted with respect to the user selected baud rate.

Regardless of the selected baud rate, the information transmitted by the GPS 25LP is referenced to the one-pulse-per-second output pulse immediately preceding the GPRMC sentence.

The GPS 25LP will transmit each sentence (except where noted in particular transmitted sentence descriptions) at a periodic rate based on the user selected baud rate and user selected output sentences. The sensor board will transmit the selected sentences contiguously. The contiguous transmission starts at a GPS second boundary. The length of the transmission can be determined by the following equation and tables:

\[
\text{length of transmission} = \frac{\text{total characters to be transmitted}}{\text{characters transmitted per sec}}
\]
<table>
<thead>
<tr>
<th>Baud</th>
<th>characters_transmitted_per_sec</th>
</tr>
</thead>
<tbody>
<tr>
<td>300</td>
<td>30</td>
</tr>
<tr>
<td>600</td>
<td>60</td>
</tr>
<tr>
<td>1200</td>
<td>120</td>
</tr>
<tr>
<td>2400</td>
<td>240</td>
</tr>
<tr>
<td>4800</td>
<td>480</td>
</tr>
<tr>
<td>9600</td>
<td>960</td>
</tr>
<tr>
<td>19200</td>
<td>1920</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>Sentence</th>
<th>max_characters</th>
</tr>
</thead>
<tbody>
<tr>
<td>GPGGA</td>
<td>72</td>
</tr>
<tr>
<td>GPGSA</td>
<td>65</td>
</tr>
<tr>
<td>GPGSV</td>
<td>210</td>
</tr>
<tr>
<td>GPRMC</td>
<td>70</td>
</tr>
<tr>
<td>GPVTG</td>
<td>34</td>
</tr>
<tr>
<td>PGRME</td>
<td>36</td>
</tr>
<tr>
<td>PGRMT</td>
<td>47</td>
</tr>
<tr>
<td>PGRMV</td>
<td>26</td>
</tr>
<tr>
<td>PGRMF</td>
<td>79</td>
</tr>
<tr>
<td>LCGLL</td>
<td>36</td>
</tr>
<tr>
<td>LCVTG</td>
<td>34</td>
</tr>
</tbody>
</table>

The factory set defaults will result in a once per second transmission at the NMEA specification transmission rate of 4800 baud.

### 4.2.2 Transmitted Time

The GPS 25LP sensor boards output UTC (Coordinated Universal Time) date and time of day in the transmitted sentences. Prior to the initial position fix, the date and time of day are provided by the on-board clock. After the initial position fix, the date and time of day are calculated using GPS satellite information and are synchronized with the one-pulse-per-second output.

The GPS 25LP uses information obtained from the GPS satellites to add or delete UTC leap seconds and correct the transmitted date and time of day. The transmitted date and time of day for leap second correction follow the guidelines in “National Institute of Standards and Technology Special Publication 432 (Revised 1990)” (for sale by the Superintendent of Documents, U.S. Government Printing Office, Washington, D.C., 20402, U.S.A.).

When a positive leap second is required, the second is inserted beginning at 23h 59m 60s of the last day of a month and ending at 0h 0m 0s of the first day of the following month. The minute containing the leap second is 61 seconds long. The GPS 25LP
would have transmitted this information for the leap second added December 31, 1989 as follows:

<table>
<thead>
<tr>
<th>Date</th>
<th>Time</th>
</tr>
</thead>
<tbody>
<tr>
<td>311289</td>
<td>235959</td>
</tr>
<tr>
<td>311289</td>
<td>235960</td>
</tr>
<tr>
<td>010190</td>
<td>000000</td>
</tr>
</tbody>
</table>

If a negative leap second should be required, one second will be deleted at the end of some UTC month. The minute containing the leap second will be only 59 seconds long. In this case, the GPS 25LP will not transmit the time of day 23h 59m 59s for the day from which the leap second is removed.

4.2.3 Global Positioning System Almanac Data (ALM)

$GPALM,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>,<10>,<11>,<12>,<13>,<14>,<15>*hh<CR><LF>

Almanac sentences are not normally transmitted. Almanac transmission can be initiated by sending the sensor board a $PGRMO,GPALM,1 command. Upon receipt of this command the sensor board will transmit available almanac information on GPALM sentences. During the transmission of almanac sentences other NMEA data output will be temporarily suspended.

<field information> can be found in section 4.1.1.

4.2.4 Global Positioning System Fix Data (GGA)

$GPGGA,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>,M,<10>,M,<11>,<12>*hh<CR><LF>

<1> UTC time of position fix, hhmmss format
<2>Latitude, dddmm.mmmm format (leading zeros will be transmitted)
<3>Latitude hemisphere, N or S
<4>Longitude, dddmm.mmmm format (leading zeros will be transmitted)
<5>Longitude hemisphere, E or W
<6>GPS quality indication, 0 = fix not available, 1 = Non-differential GPS fix available, 2 = Differential GPS (DGPS) fix available
<7>Number of satellites in use, 00 to 12 (leading zeros will be transmitted)
<8>Horizontal dilution of precision, 1.0 to 99.9
<9>Antenna height above/below mean sea level, -9999.9 to 99999.9 meters
<10>Geoidal height, -999.9 to 99999.9 meters
<11>Differential GPS (RTCM SC-104) data age, number of seconds since last valid RTCM transmission (null if non-DGPS)
Differential Reference Station ID, 0000 to 1023 (leading zeros will be transmitted, null if non-DGPS)

4.2.5 GPS DOP and Active Satellites (GSA)

$GPGSA,<1>,<2>,<3>,<3>,<3>,<3>,<3>,<3>,<3>,<4>,<5>,<6>*hh<CR><LF>

<1> Mode, M = manual, A = automatic
<2> Fix type, 1 = not available, 2 = 2D, 3 = 3D
<3> PRN number, 01 to 32, of satellite used in solution, up to 12 transmitted (leading zeros will be transmitted)
<4> Position dilution of precision, 1.0 to 99.9
<5> Horizontal dilution of precision, 1.0 to 99.9
<6> Vertical dilution of precision, 1.0 to 99.9

4.2.6 GPS Satellites in View (GSV)

$GPGSV,<1>,<2>,<3>,<4>,<5>,<6>,<7>,...<4>,<5>,<6>,<7>*hh<CR><LF>

<1> Total number of GSV sentences to be transmitted
<2> Number of current GSV sentence
<3> Total number of satellites in view, 00 to 12 (leading zeros will be transmitted)
<4> Satellite PRN number, 01 to 32 (leading zeros will be transmitted)
<5> Satellite elevation, 00 to 90 degrees (leading zeros will be transmitted)
<6> Satellite azimuth, 000 to 359 degrees, true (leading zeros will be transmitted)
<7> Signal to noise ratio (C/No) 00 to 99 dB, null when not tracking (leading zeros will be transmitted)

NOTE: Items <4>,<5>,<6> and <7> repeat for each satellite in view to a maximum of four (4) satellites per sentence. Additional satellites in view information must be sent in subsequent sentences. These fields will be null if unused.

4.2.7 Recommended Minimum Specific GPS/TRANSIT Data (RMC)

$GPRMC,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>,<10>,<11>*hh<CR><LF>

<1> UTC time of position fix, hhmmss format
<2> Status, A = Valid position, V = NAV receiver warning
<3> Latitude, ddm.mmmm format (leading zeros will be transmitted)
<4> Latitude hemisphere, N or S
<5> Longitude, dddmm.mmmm format (leading zeros will be transmitted)
<6> Longitude hemisphere, E or W
<7> Speed over ground, 0.0 to 999.9 knots
<8> Course over ground, 000.0 to 359.9 degrees, true (leading zeros will be transmitted)
<9> UTC date of position fix, ddmmyy format
Magnetic variation, 000.0 to 180.0 degrees (leading zeros will be transmitted)
Magnetic variation direction, E or W (westerly variation adds to true course)

4.2.8 Track Made Good and Ground Speed with GPS Talker ID (VTG)

The GPVTG sentence reports track and velocity information with a checksum:

\$GPVTG,<1>,T,<2>,M,<3>,N,<4>,K*hh\<CR><LF>

<1> True course over ground, 000 to 359 degrees (leading zeros will be transmitted)
<2> Magnetic course over ground, 000 to 359 degrees (leading zeros will be transmitted)
<3> Speed over ground, 00.0 to 99.9 knots (leading zeros will be transmitted)
<4> Speed over ground, 00.0 to 99.9 kilometers per hour (leading zeros will be transmitted)

4.2.9 Geographic Position with LORAN Talker ID (LCGLL)

The LCGLL sentence reports position information.

\$LCGLL,<1>,<2>,<3>,<4>,<5>,<CR><LF>

<1> Latitude, ddmm.mm format (leading zeros will be transmitted)
<2> Latitude hemisphere, N or S
<3> Longitude, dddmm.mm format (leading zeros will be transmitted)
<4> Longitude hemisphere, E or W
<5> UTC time of position fix, hhmmss format

4.2.10 Track Made Good and Ground Speed with LORAN Talker ID (LCVTG)

The LCVTG sentence reports track and velocity information.

\$LCVTG,<1>,T,<2>,M,<3>,N,<4>,K<CR><LF>

<1> True course over ground, 000 to 359 degrees (leading zeros will be transmitted)
<2> Magnetic course over ground, 000 to 359 degrees (leading zeros will be transmitted)
<3> Speed over ground, 00.0 to 99.9 knots (leading zeros will be transmitted)
<4> Speed over ground, 00.0 to 99.9 kilometers per hour (leading zeros will be transmitted)

4.2.11 Estimated Error Information (PGRME)

The GARMIN Proprietary sentence \$PGRME reports estimated position error information.
$PGRME,<1>,M,<2>,M,<3>,M*hh<CR><LF>

<1> Estimated horizontal position error (HPE), 0.0 to 9999.9 meters
<2> Estimated vertical position error (VPE), 0.0 to 9999.9 meters
<3> Estimated position error (EPE), 0.0 to 9999.9 meters

4.2.12 GPS Fix Data Sentence (PGRMF)

The sentence $PGRME is GARMIN Proprietary format.

$PGRMF,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>,<10>,<11>,<12>,<13>,<14>,<15>*hh<CR><LF>

<1> GPS week number (0 - 1023)
<2> GPS seconds (0 - 604799)
<3> UTC date of position fix, ddmmyy format
<4> UTC time of position fix, hhmmss format
<5> GPS leap second count
<6> Latitude, ddmm.mmmm format (leading zeros will be transmitted)
<7> Latitude hemisphere, N or S
<8> Longitude, dddmm.mmmm format (leading zeros will be transmitted)
<9> Longitude hemisphere, E or W
<10> Mode, M = manual, A = automatic
<11> Fix type, 0 = no fix, 1 = 2D fix, 2 = 3D fix
<12> Speed over ground, 0 to 999 kilometers/hour
<13> Course over ground, 0 to 359 degrees, true
<14> Position dilution of precision, 0 to 9 (rounded to nearest integer value)
<15> Time dilution of precision, 0 to 9 (rounded to nearest integer value)

4.2.13 Sensor Status Information (PGRMT)

The GARMIN Proprietary sentence $PGRMT gives information concerning the status of the sensor board. This sentence is transmitted once per minute regardless of the selected baud rate.

$PGRMT,<1>,<2>,<3>,<4>,<5>,<6>,<7>,<8>,<9>*hh<CR><LF>

<1> Product, model and software version (variable length field, e.g., “GPS 25LP VER 1.10”) 
<2> ROM checksum test, P = pass, F = fail 
<3> Receiver failure discrete, P = pass, F = fail 
<4> Stored data lost, R = retained, L = lost 
<5> Real time clock lost, R = retained, L = lost 
<6> Oscillator drift discrete, P = pass, F = excessive drift detected 
<7> Data collection discrete, C = collecting, null if not collecting
4.2.14 3D velocity Information (PGRMV)

The GARMIN Proprietary sentence $PGRMV reports three-dimensional velocity information.

$PGRMV,<1>,<2>,<3>*hh<CR><LF>

<1> True east velocity, -999.9 to 9999.9 meters/second
<2> True north velocity, -999.9 to 9999.9 meters/second
<3> Up velocity, -999.9 to 9999.9 meters/second

4.3 Baud Rate Selection

Baud rate selection can be performed by sending the appropriate configuration sentence to the sensor board as described in the NMEA input sentences selection.

4.4 One-Pulse-Per-Second Output

The highly accurate one-pulse-per-second output is provided for applications requiring precise timing measurements. The signal is generated after the initial position fix has been calculated and continues until power down. The rising edge of the signal is synchronized to the start of each GPS second.

Regardless of the selected baud rate, the information transmitted by the GPS 25LP sensor boards is referenced to the pulse immediately preceding the NMEA 0183 RMC sentence.

The accuracy of the one-pulse-per-second output is maintained only while the GPS 25LP can compute a valid position fix. To obtain the most accurate results, the one-pulse-per-second output should be calibrated against a local time reference to compensate for cable and internal receiver delays and the local time bias.

The default pulse width is 100 msec, however; it may be programmed in 20 msec increments between 20 msec and 980 msec as described in $PGRMC Section 4.1.3 character <13>.
4.5 RTCM Received Data

Position accuracy of less than 5 meters can be achieved with the GPS 25LP by using Differential GPS (DGPS) real-time pseudo-range correction data in RTCM SC-104 format, with message types 1, 2, 3, and 9. These corrections can be received by the GPS 25LP sensor board on RXD2. Correction data at speeds of 300, 600, 1200, 2400, 4800 or 9600 baud can be utilized, as the GPS 25LP automatically detects the incoming baud rate. For details on the SC-104 format, refer to RTCM Paper 134-89/SC 104-68 by the Radio Technical Commission for Maritime Services.
Appendix A

Earth Datums

The following is a list of the GARMIN GPS 25LP earth datum indexes and the corresponding earth datum name (including the area of application):

0  ADINDAN - Ethiopia, Mali, Senegal, Sudan
1  AFGOOYE - Somalia
2  AIN EL ABD 1970 - Bahrain Island, Saudi Arabia
3  ANNA 1 ASTRO 1965 - Cocos Island
4  ARC 1950 - Botswana, Lesotho, Malawi, Swaziland, Zaire, Zambia, Zimbabwe
5  ARC 1960 - Kenya, Tanzania
6  ASCENSION ISLAND 1958 - Ascension Island
7  ASTRO BEACON “E” - Iwo Jima Island
8  AUSTRALIAN GEODETIC 1966 - Australia, Tasmania Island
9  AUSTRALIAN GEODETIC 1984 - Australia, Tasmania Island
10  ASTRO DOS 71/4 - St. Helena Island
11  ASTRONOMIC STATION 1952 - Marcus Island
12  ASTRO B4 SOROL ATOLL - Tern Island
13  BELLEVUE (IGN) - Efate and Erromango Islands
14  BERMUDA 1957 - Bermuda Islands
15  BOGOTA OBSERVATORY - Colombia
16  CAMPO INCHAUSPE - Argentina
17  CANTON ASTRO 1966 - Phoenix Islands
18  CAPE CANAVERAL - Florida, Bahama Islands
19  CAPE - South Africa
20  CARTHAGE - Tunisia
21  CHATHAM 1971 - Chatham Island (New Zealand)
22  CHUA ASTRO - Paraguay
23  CORREGO ALEGRE - Brazil
24  DJAKARTA (BATAVIA) - Sumatra Island (Indonesia)
25  DOS 1968 - Gizo Island (New Georgia Islands)
26  EASTER ISLAND 1967 - Easter Island
27  EUROPEAN 1950 - Austria, Belgium, Denmark, Finland, France, Germany, Gibraltar, Greece, Italy, Luxembourg, Netherlands, Norway, Portugal, Spain, Sweden, Switzerland
28  EUROPEAN 1979 - Austria, Finland, Netherlands, Norway, Spain, Sweden, Switzerland
29  FINLAND HAYFORD 1910 - Finland
30  GANDAJIKA BASE - Republic of Maldives
31  GEODETIC DATUM 1949 - New Zealand
<table>
<thead>
<tr>
<th>Number</th>
<th>Description</th>
</tr>
</thead>
<tbody>
<tr>
<td>32</td>
<td>ORDNANCE SURVEY OF GREAT BRITAIN 1936 - England, Isle of Man, Scotland, Shetland Islands, Wales</td>
</tr>
<tr>
<td>33</td>
<td>GUAM 1963 - Guam Island</td>
</tr>
<tr>
<td>34</td>
<td>GUX 1 ASTRO - Guadalcanal Island</td>
</tr>
<tr>
<td>35</td>
<td>HJORSEY 1955 - Iceland</td>
</tr>
<tr>
<td>36</td>
<td>HONG KONG 1963 - Hong Kong</td>
</tr>
<tr>
<td>37</td>
<td>INDIAN - Bangladesh, India, Nepal</td>
</tr>
<tr>
<td>38</td>
<td>INDIAN - Thailand, Vietnam</td>
</tr>
<tr>
<td>39</td>
<td>IRELAND 1965 - Ireland</td>
</tr>
<tr>
<td>40</td>
<td>ISTS O73 ASTRO 1969 - Diego Garcia</td>
</tr>
<tr>
<td>41</td>
<td>JOHNSTON ISLAND 1961 - Johnston Island</td>
</tr>
<tr>
<td>42</td>
<td>KANDAWALA - Sri Lanka</td>
</tr>
<tr>
<td>43</td>
<td>KERGUELEN ISLAND - Kerguelen Island</td>
</tr>
<tr>
<td>44</td>
<td>KERTAU 1948 - West Malaysia, Singapore</td>
</tr>
<tr>
<td>45</td>
<td>L.C. 5 ASTRO - Cayman Brac Island</td>
</tr>
<tr>
<td>46</td>
<td>LGBIA 1964 - Liberia</td>
</tr>
<tr>
<td>47</td>
<td>LUZON - Mindanao Island</td>
</tr>
<tr>
<td>48</td>
<td>LUZON - Philippines (excluding Mindanao Island)</td>
</tr>
<tr>
<td>49</td>
<td>MAHE 1971 - Mahe Island</td>
</tr>
<tr>
<td>50</td>
<td>MARCO ASTRO - Salvage Islands</td>
</tr>
<tr>
<td>51</td>
<td>MASSAWA - Eritrea (Ethiopia)</td>
</tr>
<tr>
<td>52</td>
<td>MERCHICH - Morocco</td>
</tr>
<tr>
<td>53</td>
<td>MIDWAY ASTRO 1961 - Midway Island</td>
</tr>
<tr>
<td>54</td>
<td>MINNA - Nigeria</td>
</tr>
<tr>
<td>55</td>
<td>NORTH AMERICAN 1927 - Alaska</td>
</tr>
<tr>
<td>56</td>
<td>NORTH AMERICAN 1927 - Bahamas (excluding San Salvador Island)</td>
</tr>
<tr>
<td>57</td>
<td>NORTH AMERICAN 1927 - Central America (Belize, Costa Rica, El Salvador, Guatemala, Honduras, Nicaragua)</td>
</tr>
<tr>
<td>58</td>
<td>NORTH AMERICAN 1927 - Canal Zone</td>
</tr>
<tr>
<td>59</td>
<td>NORTH AMERICAN 1927 - Canada (including Newfoundland Island)</td>
</tr>
<tr>
<td>60</td>
<td>NORTH AMERICAN 1927 - Caribbean (Barbados, Caicos Islands, Cuba, Dominican Republic, Grand Cayman, Jamaica, Leeward Islands, Turks Islands)</td>
</tr>
<tr>
<td>61</td>
<td>NORTH AMERICAN 1927 - Mean Value (CONUS)</td>
</tr>
<tr>
<td>62</td>
<td>NORTH AMERICAN 1927 - Cuba</td>
</tr>
<tr>
<td>63</td>
<td>NORTH AMERICAN 1927 - Greenland (Hayes Peninsula)</td>
</tr>
<tr>
<td>64</td>
<td>NORTH AMERICAN 1927 - Mexico</td>
</tr>
<tr>
<td>65</td>
<td>NORTH AMERICAN 1927 - San Salvador Island</td>
</tr>
<tr>
<td>66</td>
<td>NORTH AMERICAN 1983 - Alaska, Canada, Central America, CONUS, Mexico</td>
</tr>
<tr>
<td>67</td>
<td>NAPARIMA, BWI - Trinidad and Tobago</td>
</tr>
<tr>
<td>68</td>
<td>NAHRWAN - Masirah Island (Oman)</td>
</tr>
<tr>
<td>69</td>
<td>NAHRWAN - Saudi Arabia</td>
</tr>
<tr>
<td>70</td>
<td>NAHRWAN - United Arab Emirates</td>
</tr>
<tr>
<td>71</td>
<td>OBSERVATORIO 1966 - Corvo and Flores Islands (Azores)</td>
</tr>
<tr>
<td></td>
<td></td>
</tr>
<tr>
<td>---</td>
<td>----------</td>
</tr>
<tr>
<td>72</td>
<td>OLD EGYPTIAN - Egypt</td>
</tr>
<tr>
<td>73</td>
<td>OLD HAWAIIAN - Mean Value</td>
</tr>
<tr>
<td>74</td>
<td>OMAN - Oman</td>
</tr>
<tr>
<td>75</td>
<td>PICO DE LAS NIEVES - Canary Islands</td>
</tr>
<tr>
<td>76</td>
<td>PITCAIRN ASTRO 1967 - Pitcairn Island</td>
</tr>
<tr>
<td>77</td>
<td>PUERTO RICO - Puerto Rico, Virgin Islands</td>
</tr>
<tr>
<td>78</td>
<td>QATAR NATIONAL - Qatar</td>
</tr>
<tr>
<td>79</td>
<td>QORNOQ - South Greenland</td>
</tr>
<tr>
<td>80</td>
<td>REUNION - Mascarene Island</td>
</tr>
<tr>
<td>81</td>
<td>ROME 1940 - Sardinia Island</td>
</tr>
<tr>
<td>82</td>
<td>RT 90 - Sweden</td>
</tr>
<tr>
<td>83</td>
<td>PROVISIONAL SOUTH AMERICAN 1956 - Bolivia, Chile, Colombia, Ecuador, Guyana, Peru, Venezuela</td>
</tr>
<tr>
<td>84</td>
<td>SOUTH AMERICAN 1969 - Argentina, Bolivia, Brazil, Chile, Colombia, Ecuador, Guyana, Paraguay, Peru, Venezuela, Trinidad and Tobago</td>
</tr>
<tr>
<td>85</td>
<td>SOUTH ASIA - Singapore</td>
</tr>
<tr>
<td>86</td>
<td>PROVISIONAL SOUTH CHILEAN 1963 - South Chile</td>
</tr>
<tr>
<td>87</td>
<td>SANTO (DOS) - Espirito Santo Island</td>
</tr>
<tr>
<td>88</td>
<td>SAO BRAZ - Sao Miguel, Santa Maria Islands (Azores)</td>
</tr>
<tr>
<td>89</td>
<td>SAPPER HILL 1943 - East Falkland Island</td>
</tr>
<tr>
<td>90</td>
<td>SCHWARZECK - Namibia</td>
</tr>
<tr>
<td>91</td>
<td>SOUTHEAST BASE - Porto Santo and Madeira Islands</td>
</tr>
<tr>
<td>92</td>
<td>SOUTHWEST BASE - Faial, Graciosa, Pico, Sao Jorge, and Terceira Islands (Azores)</td>
</tr>
<tr>
<td>93</td>
<td>TIMBALAI 1948 - Brunei and East Malaysia (Sarawak and Sabah)</td>
</tr>
<tr>
<td>94</td>
<td>TOKYO - Japan, Korea, Okinawa</td>
</tr>
<tr>
<td>95</td>
<td>TRISTAN ASTRO 1968 - Tristan da Cunha</td>
</tr>
<tr>
<td>96</td>
<td>User defined earth datum</td>
</tr>
<tr>
<td>97</td>
<td>VITI LEVU 1916 - Viti Levu Island (Fiji Islands)</td>
</tr>
<tr>
<td>98</td>
<td>WAKE-ENIWETOK 1960 - Marshall Islands</td>
</tr>
<tr>
<td>99</td>
<td>WORLD GEODETIC SYSTEM 1972</td>
</tr>
<tr>
<td>100</td>
<td>WORLD GEODETIC SYSTEM 1984</td>
</tr>
<tr>
<td>101</td>
<td>ZANDERIJ - Surinam</td>
</tr>
<tr>
<td>102</td>
<td>CH-1903 - Switzerland</td>
</tr>
<tr>
<td>103</td>
<td>Hu - Tzu - Shan</td>
</tr>
<tr>
<td>104</td>
<td>Indonesia 74</td>
</tr>
<tr>
<td>105</td>
<td>Austria</td>
</tr>
<tr>
<td>106</td>
<td>Potsdam</td>
</tr>
<tr>
<td>107</td>
<td>Taiwan (modified Hu-Tzu-Shan)</td>
</tr>
</tbody>
</table>
Appendix B

GPS 25LP Connectors

The mating power/data connector is available from JST Corporation, you may contact them for the current distributor nearest you. The information is listed below.

Mating Connector Part Number:

<table>
<thead>
<tr>
<th>Housing:</th>
<th>ZHR-12</th>
</tr>
</thead>
<tbody>
<tr>
<td>Crimp Contact:</td>
<td>SZH-002T-P0.5 (for 26 to 28 AWG wire)</td>
</tr>
<tr>
<td></td>
<td>SZH-003T-P0.5 (for 28 to 32 AWG wire)</td>
</tr>
</tbody>
</table>

**U.S.A.**

J.S.T. CORPORATION
1957 South Lakeside Drive
Waukegan, IL, 60085 U.S.A.
Phone: (847) 473-1957
Fax<4>: (847) 473-0144

**JAPAN**

J.S.T. MFG CO., LTD.
6-8, 2-chome,
Shigino -nishi
Joto-ku
Osaka 536, JAPAN
Phone: (06) 968-6855
Fax<4>: (06) 964-2085

A completely assembled connector with twelve inch pre-stripped wires is available from GARMIN Corporation as part number 325-00026-00. Contact GARMIN for pricing and availability.

An MCX terminated RF cable pig-tail assembly is also available from GARMIN Corporation as part number 320-00048-00.
Appendix C

GPS 25LP Evaluation Kits

GPS 25LP evaluation kits, part numbers 010-10186-00, -02, or -03, are available from GARMIN International. These kits include respectively a GPS 25-LVC, -LVS, or -HVS sensor board, a GPS antenna, a 5V or 12V power supply, an interface cable for PC serial port connection, and software to monitor GPS 25LP output and configure the sensor board.

To install the evaluation and configuration software run the program, setup, by using the FILE - RUN command in Windows.

The NMEAVWR.EXE program in the GARMIN program group can accept NMEA data from either the com1 or com2 PC serial port at 1200, 2400, 4800, 9600 or 19200 baud. The default settings are com1 at 4800 baud. NMEAVWR receives NMEA sentences and displays sentence information in a formatted display on the screen. The top portion of the screen displays the 5 character identifier of sentences received, the age in seconds since the last transmission, and a count of the number of times the sentence has been received. The middle portion of the screen displays the most recently received data in the sentence with the selected NMEA identifier (highlighted in the top portion of the screen using the arrow keys). The lower portion contains a formatted presentation of the currently selected sentence. In addition to receiving data the program will also upload NMEA sentences to the sensor board. The ALT-U key sequence will upload the file NMEA.TXT in the current directory to the unit. Received data can also be logged to a file. NMEAVWR can be invoked from a DOS prompt with the following optional parameters:

```
<path>nmeavwr [/b:<baud>] [/<port>] [log_file.txt]
```

where:

[] brackets indicate optional parameters

- `<path>` DOS path to nmeavwr.exe if not in current directory
- `<baud>` baud rate
- `<port>` PC communications port
- `log_file.txt` ASCII log file of all received sentences

Example:
```
c:\garmin>nmeavwr /b:9600 /com2 log_file.txt
```

If no options are used, the defaults are 4800 baud, com1, and no data logging.

The configuration program GPSCFG.EXE will configure the sensor boards based on user selected parameters. Some program features include the ability to download sensor board configuration, maintain different configurations in files, and perform sensor board configurations quickly with the use of one function key. Online program help is available.